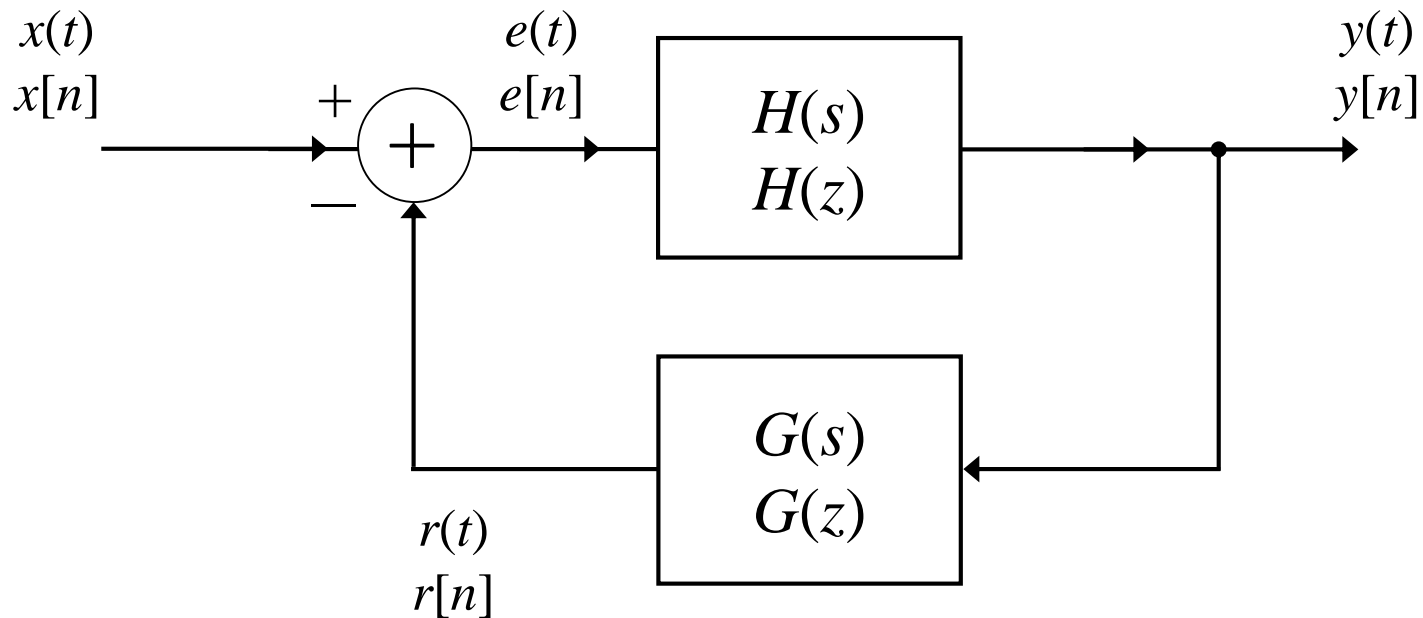


# 11.0 Linear Feedback Systems

## *11.1 Basic Principles of Linear Feedback*



# 11.0 Linear Feedback Systems

## *11.1 Basic Principles of Linear Feedback*

$H(s), H(z)$ : forward path system function

$G(s), G(z)$ : feedback path system function

$Q(s), Q(z)$ : close-loop system function

$$Q(s) = \frac{H(s)}{1 + G(s)H(s)}, \quad Q(z) = \frac{H(z)}{1 + G(z)H(z)}$$

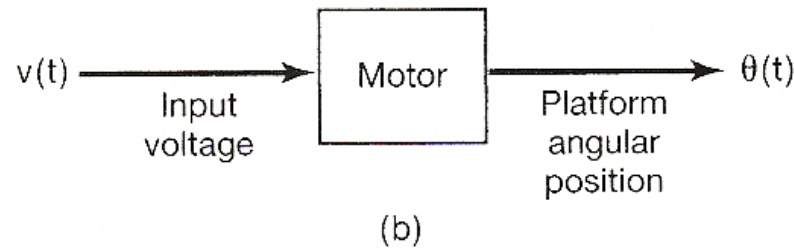
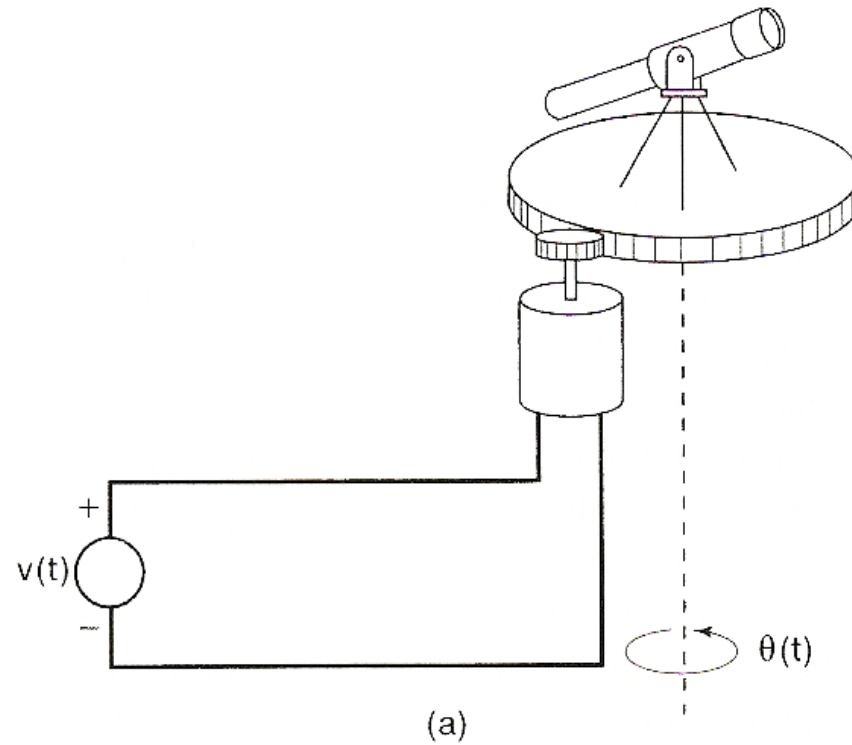
## *Example 1: telescope positioning system*

*See Fig. 11.1, p.817 of text*

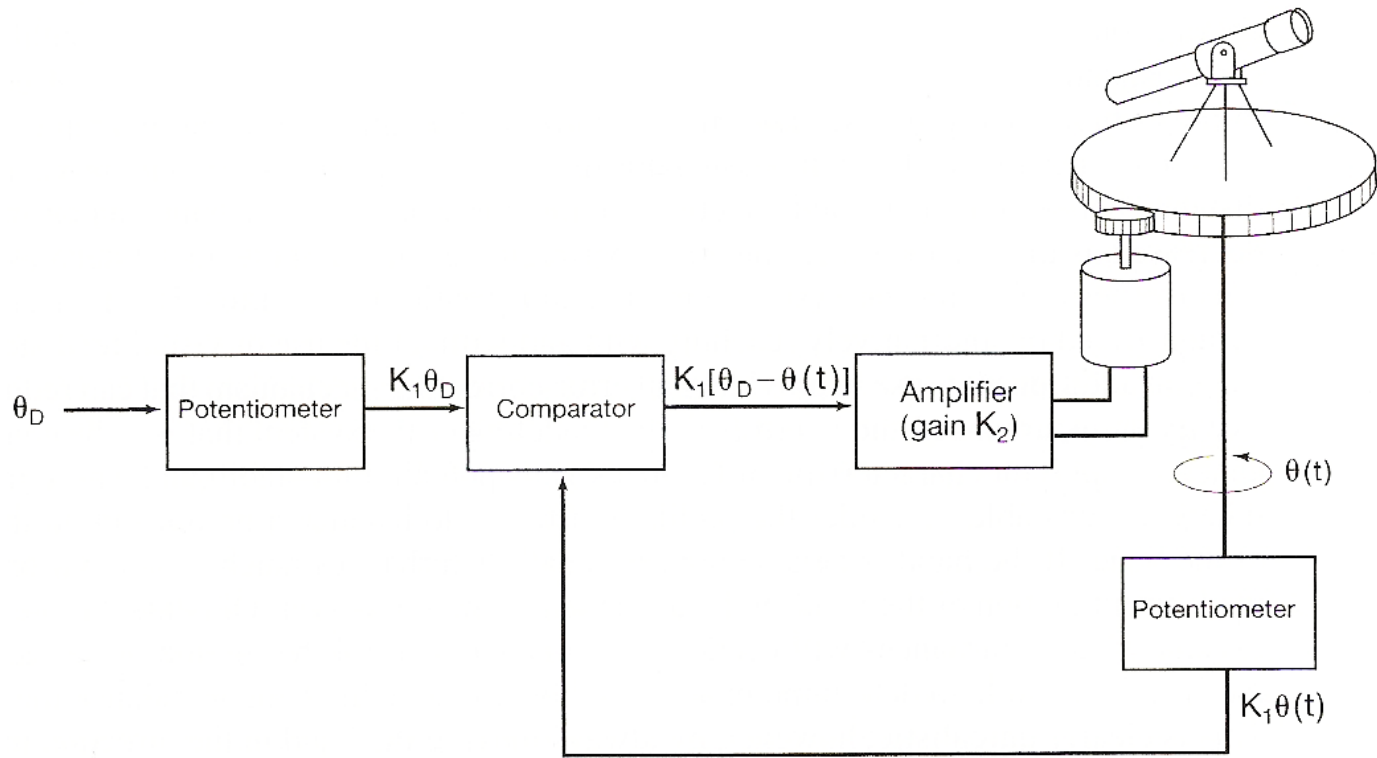
close-loop system vs. open-loop system

- insensitivity to precise knowledge of the open-loop system characteristics
- insensitivity to disturbances
- tolerance to errors

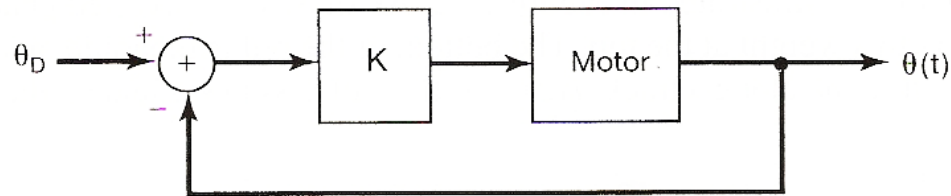
## *Example 2: autopilot system for airplanes*



**Figure 11.1** Use of feedback to control the angular position of a telescope: (a) dc motor-driven telescope platform; (b) block diagram of the system in (a); (c) feedback system for pointing the telescope; (d) block diagram of the system in (c) (here,  $K = K_1 K_2$ ).



(c)



(d)

**Figure 11.1** Use of feedback to control the angular position of a telescope: (a) dc motor-driven telescope platform; (b) block diagram of the system in (a); (c) feedback system for pointing the telescope; (d) block diagram of the system in (c) (here,  $K = K_1 K_2$ ).

- Stabilization of Unstable Systems

Example 3:

$$H(s) = \frac{b}{s - a}, \text{ unstable if } a > 0$$

$$Q(s) = \frac{H(s)}{1 + KH(s)} = \frac{b}{s - a + Kb}$$

$$\text{stable if } K > \frac{a}{b}$$

- Stabilization of Unstable Systems

Example 4:

$$H(s) = \frac{b}{s^2 + a}, \text{ unstable}$$

$$G(s) = K_1 + K_2s$$

stable if  $bK_2 > 0$ ,  $a + K_1b > 0$

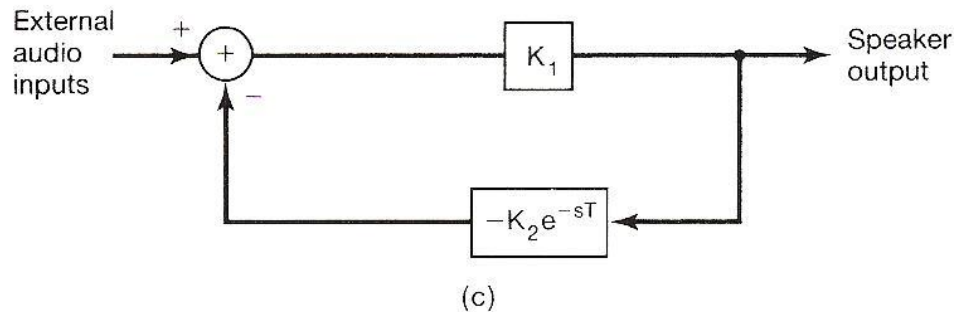
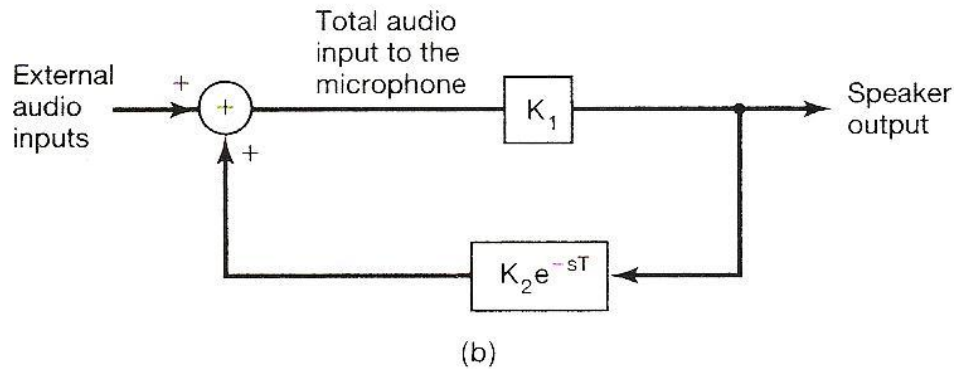
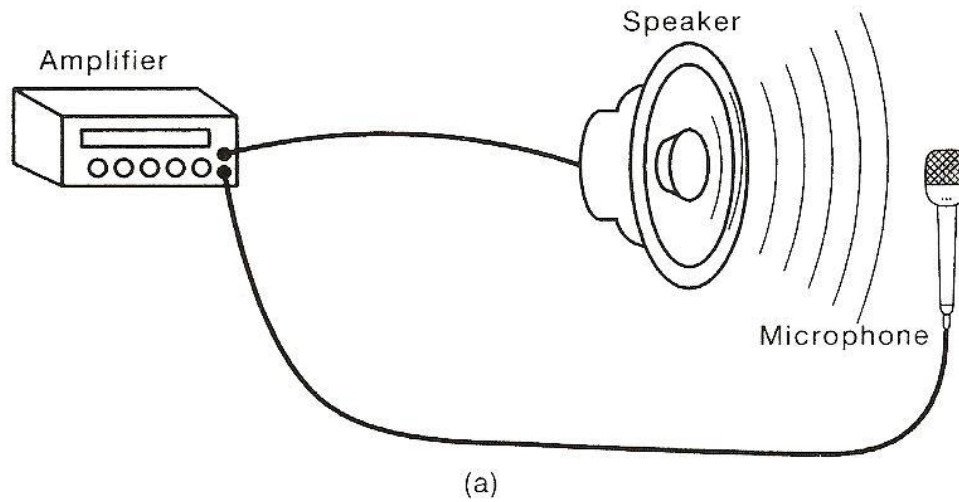
- Destablization Caused by Feedback

*Example 5: audio feedback*

*See Fig. 11.8, p.831 of text*

unstable if  $K_1K_2 > 1$





**Figure 11.8** (a) Pictorial representation of the phenomenon of audio feedback; (b) block diagram representation of (a); (c) block diagram in (b) redrawn as a negative feedback system. (Note:  $e^{-sT}$  is the system function of a  $T$ -second time delay.)

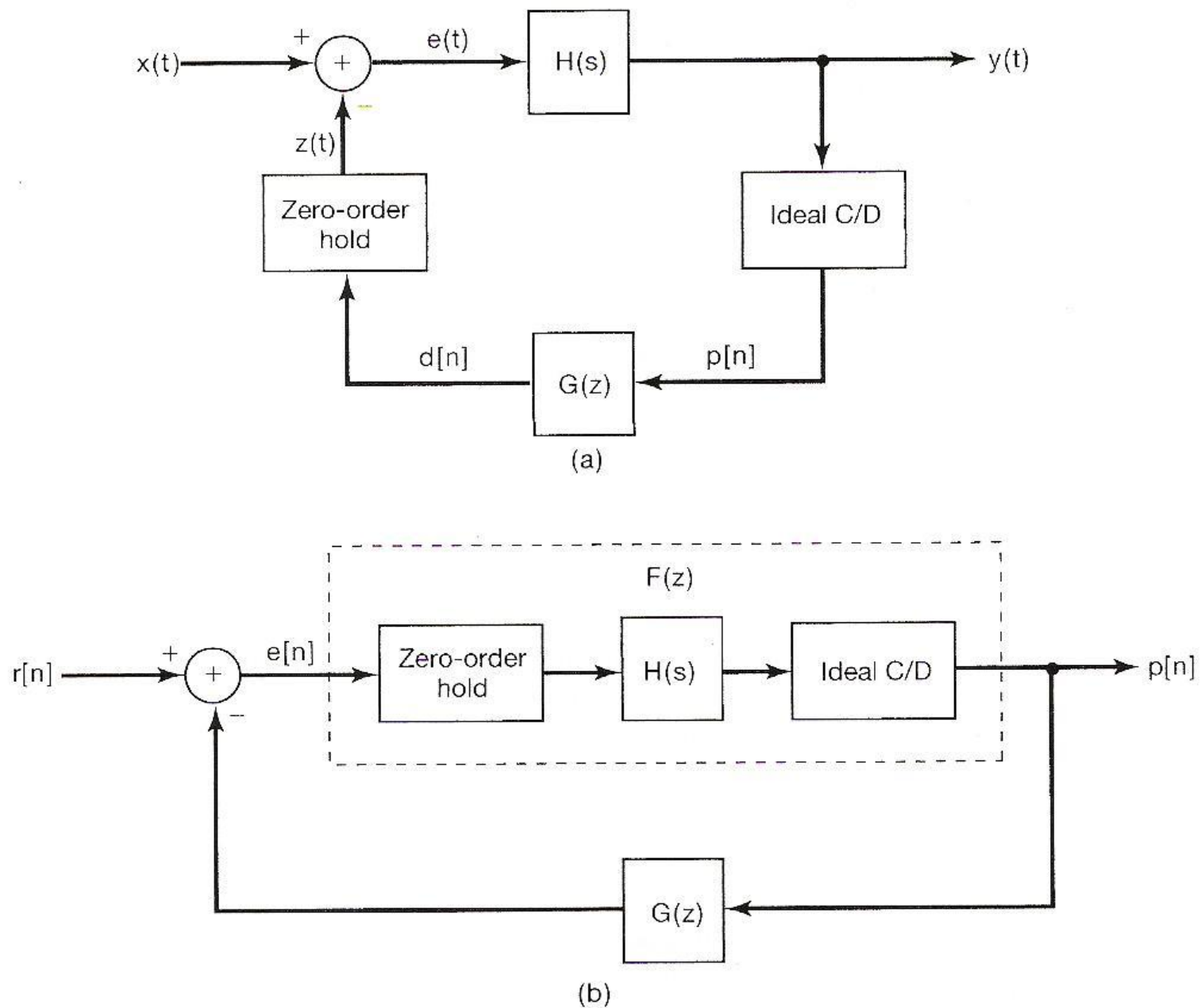
- **Sampled-Data Feedback System**

discrete-time feedback for continuous-time systems

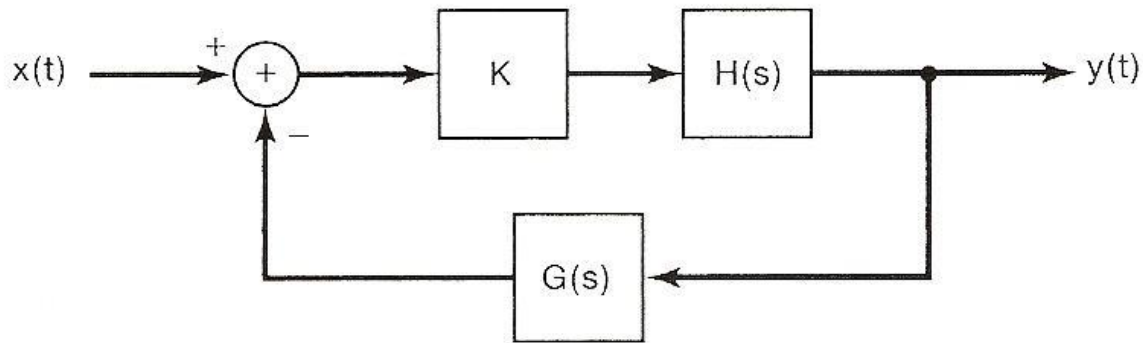
*See Fig. 11.6, p.827 of text*

- **The Adjustable Gain can be on either arm**

*See Fig. 11.10, p.835 of text*

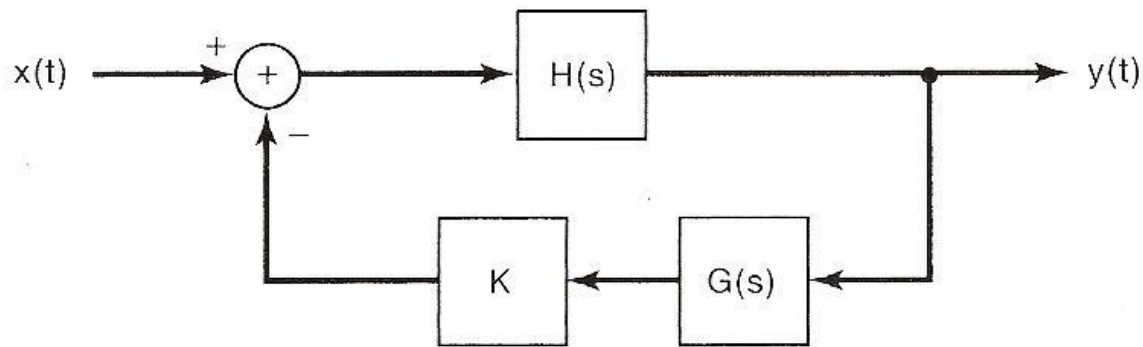


**Figure 11.6** (a) A sampled-data feedback system using a zero-order hold; (b) equivalent discrete-time system.



$$Q(s) = \frac{KH(s)}{1 + KH(s)G(s)}$$

(a)



$$Q(s) = \frac{H(s)}{1 + KH(s)G(s)}$$

(b)

**Figure 11.10** Feedback systems containing an adjustable gain: (a) system in which the gain is located in the forward path; (b) system with the gain in the feedback path.

## ● Root-locus Analysis for Feedback Systems

- the locus (path) for the poles of the close-loop system when an adjustable gain is varied

### Example 6:

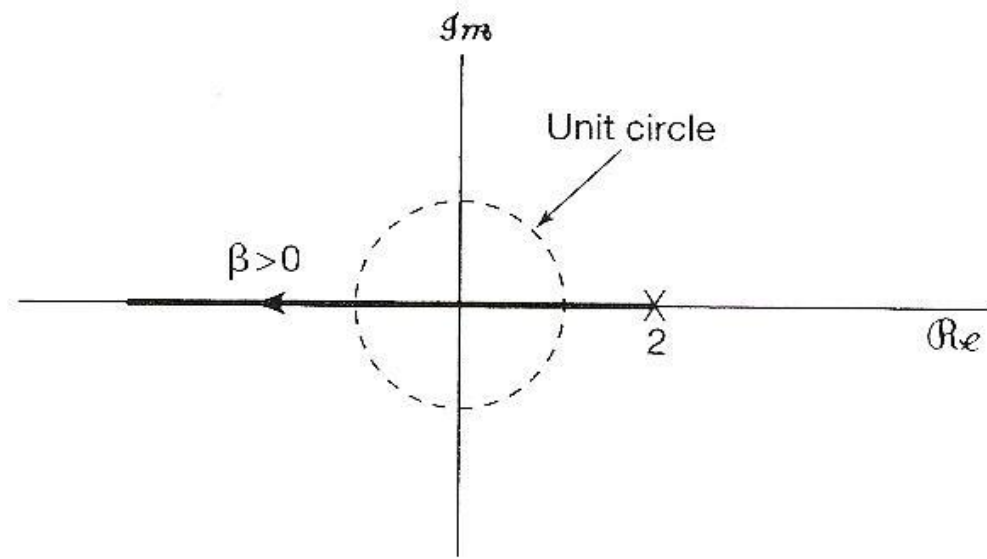
$$H(z) = \frac{1}{1 - 2z^{-1}} = \frac{z}{z - 2}$$

$$G(z) = 2\beta z^{-1} = \frac{2\beta}{z}$$

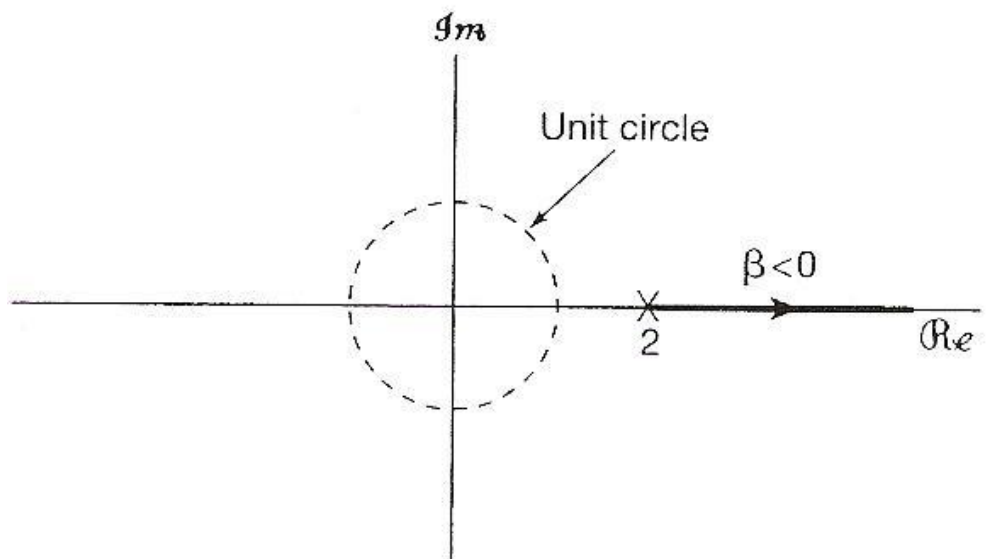
$$Q(z) = \frac{1}{1 - 2(1 - \beta)z^{-1}} = \frac{z}{z - 2(1 - \beta)}$$

pole at  $z = 2(1 - \beta)$

*See Fig. 11.9, p.834 of text*



(a)



(b)

**Figure 11.9** Root locus for the closed-loop system of eq. (11.40) as the value of  $\beta$  is varied: (a)  $\beta > 0$ ; (b)  $\beta < 0$ . Note that we have marked the point  $z = 2$  that corresponds to the pole location when  $\beta = 0$ .

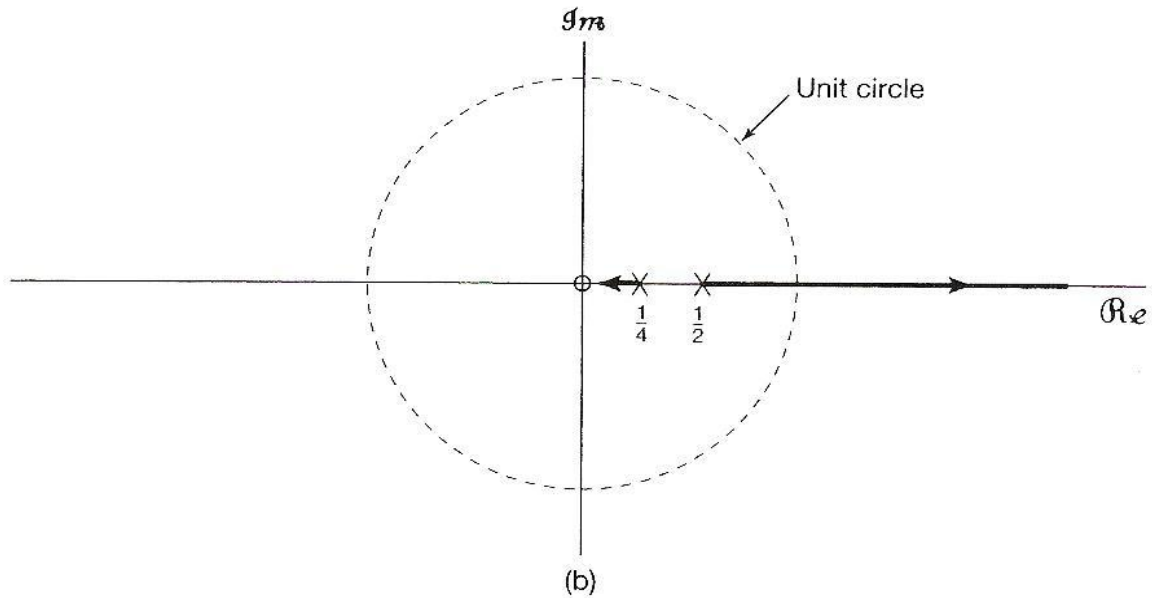
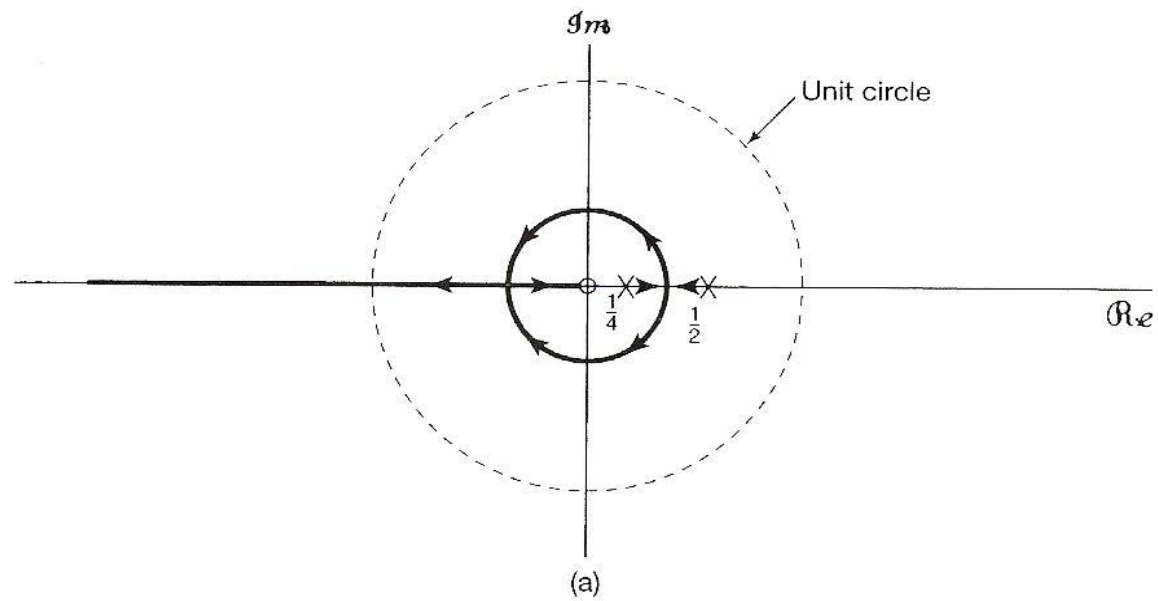
## ● Root-locus Analysis for Feedback Systems

- the locus (path) for the poles of the close-loop system when an adjustable gain is varied

Example 7:

$$H(z) = \frac{K}{1 - \frac{1}{2}z^{-1}}, \quad G(z) = \frac{z^{-1}}{z - \frac{1}{4}z^{-1}}$$

*See Fig. 11.16, p.846 of text*



**Figure 11.16** Root locus for Example 11.3: (a)  $K > 0$ ; (b)  $K < 0$ . The poles of  $G(z)H(z)$  at  $z = 1/4$  and  $z = 1/2$  and the zero of  $G(z)H(z)$  at  $z = 0$  are indicated in the figure.



***END Of THIS COURSE***